



SERVICE BULLETIN

FEBRUARY 8, 2023

MORE COMPANY S.B. NO. 103

Rev. 0

PT6A TURBOPROP ENGINE
ELIMINATE TEMPORARY CT DISK
INSPECTION REQUIREMENTS

1. Planning Information
 - A. Effectivity

Applies to PRATT & WHITNEY CANADA PT6A engines using the following MORE
STC: SE00001EN PT6A-41, -42, -42A
 - B. Concurrent Requirements

Resume the standard practice of inspecting the Compressor Turbine Disk via
Borescope at 400/450 hour intervals.
 - C. Purpose

To eliminate the temporary requirement for **Compressor Turbine Disk Inspection**
to "split" the engine and visually inspect the Disk for cracks in the area not
viewable with a Borescope. Compliance with paragraph E1. pg 46 of MORE
STC manual, is no longer required.
 - D. Description

MORE STC SE00001EN contains a temporary work task at 400/450 hour
intervals, to inspect the Compressor Turbine Disk for rim cracks by "splitting" the
engine and visually inspecting the Disk for cracks in the area not viewable with
a Borescope. A review of the cracking issue over the past twenty years
indicates this practice is no longer necessary for P&WC PT6A-41,
-42, -42A engines.
 - E. Reference

P&WC S/B 3136, P&WC S/B 3149, P&WC S/B 3164, P&WC S/B 3187, P&WC S/B
3301, P&WC S/B 3318, P&WC S/B 3332, P&WC S.I.L. GEN-037, P&WC S.I.L.
GEN-034, P&WC S.I.L. PT6A-066, and P&WC S/B 3360 indicated that the
compressor turbine disk rim failure was likely caused by overheating of the
compressor turbine disk rim and these documents demonstrated P&WC's
efforts to address the problem.
 - F. Compliance

Voluntary
 - G. Approval

Federal Aviation Administration has reviewed and approved the technical
contents of this Service Bulletin, Revision Original.